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Quantifying the Effect of Macrozones on the Cold-Dwell Fatigue Response of UD-Rolled Ti-6Al-4V using High-Energy X-Ray Diffraction

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Abstract

High-energy X-ray diffraction is used to investigate the evolution of elastic lattice strains in rolled Ti-6Al-4V specimens during cyclic loading in-situ with and without the inclusion of a 120s dwell period. A 1 mm segment of the gauge section is monitored throughout the first 100 cycles for specimens extracted along the rolling direction (RD), the transverse direction (TD) and 45° between the two, to explore the effects of texture on the evolution of the micromechanical response. Five families of lattice planes from the hexagonal α phase are analysed with emphasis on lattice strain measured at the peak of each cycle, while macroscopic strain accumulation is simultaneously monitored via Digital Image Correlation. In cyclic loading conditions including a dwell period at load, a prominent increase in elastic strains is observed in prismatic and basal lattice planes with the specimen loaded 45° from the rolling direction. In the absence of dwell, both RD and TD specimen orientations exhibited subtle cyclic hardening in all families of lattice planes probed despite negligible evolution in accumulated macroscopic plastic strain. Estimations of lattice orientation-dependent stresses are also presented using directional moduli to examine redistribution of load across sets of grains with the increasing cycle count.

Keywords: Titanium, Fatigue, Cold-Dwell, Diffraction, Lattice Strain

1. Introduction

Since their introduction as commercial materials in the late 1940s, titanium alloys have become established as the primary material of choice for a variety of demanding applications in the aerospace sector, jet engines in particular. Titanium alloys generally exhibit very good specific strength, good corrosion resistance and, in some cases, very good fatigue strength [1]. Nonetheless, the deleterious effects of prolonged exposure to sustained and repeated, sub-yield load on different titanium

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8 alloys have been the subject of research for decades due to the complex interactions
9 between microstructural features that drive fatigue failure. Notably, the presence of
10 an extended dwell period at peak load (conditions encountered during the cruising
11 phase of a flight cycle) has been shown to negatively affect many titanium alloys,
12 producing a significant reduction in fatigue life [2–7]. Sensitivity to these load-
13 ing conditions, termed cold dwell fatigue, is dependent on the complex interactions
14 between composition, microstructure and processing history. For example, coarse
15 microstructures of Ti-6Al-4V (Ti64) and, in particular, those treated above the β
16 transus and then cooled slowly, have exhibited higher sensitivity to cold dwell [8].
17 However, fine-grained microstructures such as fully lamellar or Widmanstätten types
18 of microstructures have shown insensitivity to the dwell phenomenon [9]. This be-
19 haviour has led to large factors of safety being used in the design of critical parts,
20 resulting in heavier jet engine components requiring more regular inspections. A
21 significant body of literature has emerged proposing numerous contributing factors
22 to dwell fatigue failures. The need for in-situ experiments in which one can ‘observe’
23 the microstructural and micromechanical evolution driving the failure is motivated
24 by the attempt to relate multiple contributing factors to the existing theory.

25 Macroscopically, dwell failures differentiate themselves from ordinary fatigue or creep
26 failures as they are associated with the presence of quasi-cleavage facets which nu-
27 cleate beneath the surface and not necessarily at high-stress concentration points
28 [3]. At the grain scale, it is generally accepted that the stress redistribution between
29 ‘soft’ and ‘hard’ grains is driving the facet formation and subsequently the premature
30 failure. A grain is characterised as ‘soft’ when its crystallographic c axis is oriented
31 relative to the loading axis in such a way that makes crystallographic slip (or simply
32 slip) on prismatic or basal slip systems favourable. Conversely, a grain with its c
33 axis oriented parallel or near-parallel to the loading axis is described as ‘hard’ as
34 slip cannot easily occur. The dwell period allows for time-sensitive accumulation
35 of dislocations resulting from plasticity, particularly in areas where there is low slip
36 transmission, which subsequently increases the stress at the ‘hard’ grain boundary
37 [3, 10]. It has been shown that grains with a c axis deviation of up to 15° from the
38 loading axis can be characterised as ‘hard’ and constitute potential sites for facet
39 formation on failure slip planes that lie perpendicular to the loading axis [11, 12].

40 Even though basal planes have been widely regarded as the prime locations for facet
41 formation [3], Sackett et al [13] found that basal slip had not played a key role in the
42 dwell fatigue failures of Ti685 that was heat-treated to result in abnormally large
43 grains. Single-crystal, micro-pillar experiments and crystal plasticity modelling us-
44 ing Ti6242 by Zhang et al [14] showed that even though the strain-rate sensitivity
45 of basal planes is higher than that for prisms, the difference between them is smaller
46 than had been previously reported. High cycle fatigue (HCF) experiments by Ban-
47 tounas et al [15] using Ti64 plate indicated that macrozones with their c axis close to
48 the loading direction displayed facets at the fracture surface while those perpendic-
49 ular to the loading axis were proposed to have acted as barriers to slip that displayed
50 irregular fracture morphologies.

51 As lattice orientation clearly plays a critical role in the dwell fatigue process, texture

52 and macrozones (large regions of grains with similar orientation) have a significant
53 effect on failure initiation and propagation during cyclic loading. A comprehensive
54 study by Le Biavant et al [16] indicated that cracks in textured Ti64 initiated on
55 basal or prismatic planes, with a macrozone that had an orientation favourable for
56 slip appearing heavily damaged after being fatigued. Hémery et al [17] subjected
57 Ti64 to monotonic, sub-yield loading which revealed that slip activated earlier in
58 microtextured regions with strong basal texture and that basal slip was found to
59 activate before prismatic slip. Even though the orientation of the macrozone is
60 directly linked to the crack initiation in α grains, it does not affect crack growth
61 characteristics once the crack length exceeds the size of the macrozone [16]. A study
62 by Zhang et al [18] supports the case for crack initiation taking place at the interface
63 between macrozones, one of which being more and one being less favourably oriented
64 for slip. A model by Pilchak [19] showed that in the presence of dwell, the size of
65 the microtextured regions (MTRs) in titanium alloys has a much larger effect on
66 fatigue life than the initial crack size.

67 In this context, the role of the deformation mode during raw material processing
68 should be acknowledged as a key determinant of texture symmetry and intensity.
69 At relatively low deformation temperatures, unidirectionally-rolling Ti64 will form a
70 basal/transverse texture; as the rolling temperature approaches the β transus a pure
71 transverse texture can be obtained, while processing above the transus can result in
72 more intricate textures which could also include grains with their c axis along the
73 Rolling Direction (RD) [1]. More specifically, in commercially rolled Ti64 products,
74 most grains are found to preferentially orient themselves so that their c axes are
75 perpendicular to the rolling direction [20]. However, Bantounas et al [21] noted
76 the presence of two macrozone categories in UD-rolled Ti64; one with the grains' c
77 axes lying perpendicular to the RD, and one with the c axes lying parallel to the
78 RD with the former occupying a larger proportion of the scanned area. It is also
79 worth mentioning that data by Bache & Evans [20] on UD-rolled Ti64 suggests that
80 even though specimens cut along the transverse direction (TD) exhibited a superior
81 modulus of elasticity, yield stress and UTS compared to specimens along the RD,
82 this was largely offset by poor ductility.

83 There is currently limited direct experimental validation of the dwell fatigue failure
84 mechanisms due to the difficulty in bridging the gap between microscale experi-
85 ments, that do not necessarily represent the bulk behaviour and macroscale experi-
86 ments that cannot be used to reliably tie the macroscopic response to the grain-
87 level interactions beyond post-event examinations. Synchrotron diffraction offers
88 the advantage of being able to examine grain-scale response through measurement
89 of lattice strains in-situ while probing a statistically significant number of grains. In
90 the last decade, synchrotron diffraction has been used more widely to characterise
91 the behaviour of titanium alloys, including of forged Ti64 under tensile loading [22],
92 mill-annealed Ti64 [23], CP Ti [24, 25], Ti-7Al [26] and equiaxed Ti64 [27, 28].
93 While some of the previous work pays attention to stress relaxation effects during
94 extended hold periods in CP Ti [25], there are no studies that examine the lattice
95 strain response of dwell fatigue over a sustained number of cycles.

96 In order to elucidate the aforementioned mechanisms and begin to determine the role
 97 of the processing route, the evolution of lattice strains of unidirectionally-rolled (UD)
 98 Ti64 in dwell and non-dwell cyclic loading conditions is tracked in-situ using high-
 99 energy X-ray diffraction. Due to the strong MTR presence along a single direction,
 100 the UD-rolled product was chosen to impose distinctly different conditions for slip
 101 favourability by extracting specimens along the RD, TD and 45° between the two.
 102 Further information about the material and the microtexture present is provided in
 103 Section 2, as well as a detailed overview of the experimental setup. The macroscopic
 104 strain, lattice strain and estimated lattice stress results are provided in Section 3.
 105 This is followed by a comprehensive discussion.

106 2. Experiments

107 2.1. Material

108 The material studied is UD-rolled Ti64 plate supplied by Rolls-Royce plc. The ingot
 109 was initially β worked then α/β worked in an open die forge to produce the inter-
 110 mediate slab and finally α/β rolled in one direction gradually reducing its thickness.
 111 Lastly, the material was creep flattened and machined to the finished product thick-
 112 ness of 10 mm. The microstructure of the as-received plate was examined using a
 113 Zeiss Sigma SEM. Two representative images are shown in Fig. 1 which reveal a
 114 bimodal microstructure typical for a hot-rolled Ti64 product. This consists of pri-
 115 mary α phase grains which appear both as near-equiaxed with the width ranging
 116 from 5 μm to 20 μm , as well as grains with extensive elongation along the TD/RD
 117 plane, as shown in Fig. 1(b). Some of these fairly homogeneous α phase regions have
 118 been found to measure up to 110 μm , while the height of such grains does not tend to
 119 exceed 10 μm along the normal direction (ND). A smaller proportion of secondary
 120 α grains is also observed with individual α_s lamellae measuring 1-2 μm in thick-
 121 ness. The β phase is concentrated along grain boundaries, as well as in the form of
 122 finer laths within secondary alpha grains. An energy-dispersive X-ray spectroscopy
 123 (EDS) analysis was carried out on the same as-received specimen at a magnification
 124 of 3000x and a pixel size of 0.1 μm covering an area of 103 μm x 77 μm . The results
 125 of this analysis, along with the supplier's nominal composition of the plate product
 126 are summarised in Table 1. Finally, four SEM images were processed to estimate
 127 the area fraction of the β phase, which was found to be 5.01% on average.

Table 1: Nominal and measured composition of the UD-rolled Ti64 plate.

	Ti	Al	V	N	C	O	Fe	H	Total Resid.
Nominal	Balance	5.5-6.7	3.5-4.5	0.05	0.08	0.20	0.40	0.015	0.4
EDS Sample	Balance	6.0	4.0						

128 An FEI Quanta 200F was used for all EBSD scans of the raw plate material and the
 129 dwell specimens, as shown in Fig. 2(a) and 3, respectively. In the large area EBSD
 130 map, orientations are coloured relative to the transverse direction (TD) as is common
 131 for rolled products, such that red indicates a basal normal pointing along the TD.
 132 The large area scan shown in Fig. 2(a) consists of 21 tiles at 230x magnification

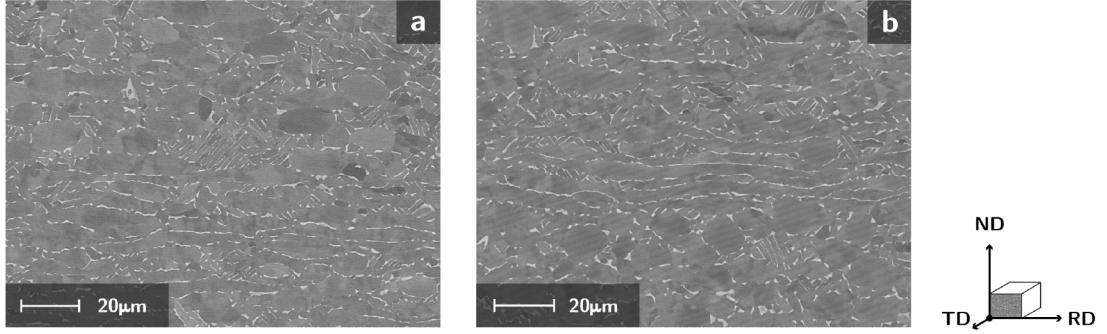


Figure 1: Angle selective Backscatter (AsB) images of the as-received material highlighting two distinct microstructural regions. (a) shows near equiaxed primary α grains (dark grey regions) with a smaller proportion of the β phase (light grey regions) concentrated along grain boundaries. Secondary α grains are also found consisting of smaller α_s lamellae and finer β . (b) shows a different region of the material where the elongation of α grains is prominent due to the effect of rolling.

133 and a step size of $3.5\ \mu\text{m}$. The synchrotron specimens covered an area of $<0.9\ \text{mm}^2$
 134 and a step size of $2\ \mu\text{m}$ was used to scan the entire cross-sectional area at the gauge
 135 in a single scan at a magnification of 110x. Tiles were manually stitched using the
 136 OIM Analysis [29] software and subsequent processing of all the EBSD scans and
 137 visualisation was carried out using a custom MTEX routine [30]. Fig. 2(b) shows
 138 the (0001) , $(10\bar{1}0)$, and $(11\bar{2}0)$ orientation pole figures to visualise the calculated
 139 orientation density function (ODF) (i.e. kernel density estimation) from the same
 140 region probed using EBSD.

141 The scans reveal a strong texture, primarily consisting of α grains with their crys-
 142 tallographic c axis along the TD and, to a much lesser extent, along the RD which
 143 is also clearly seen in the pole figures in Fig. 2(b). This is a typical texture for
 144 the UD-rolled plate [31]. The large area scan in Fig. 2(a) shows that macrozones
 145 are prevalent in the scanned area (large regions with the $[0001]$ direction aligned
 146 with TD), and highlights the presence of macrozones with widths that extend up to
 147 approximately 1 mm to 2 mm. This is an important consideration for the specimens
 148 used as the gauge diameter does not exceed 1 mm.

149 Specimens were manufactured from the rolled plate using Electrical Discharge Ma-
 150 chining (EDM) along three orientations; the RD, TD and 45° between the two.
 151 Taking the load frame capability and the X-ray beam size into account, it was de-
 152 termined that the specimen width at the gauge should not exceed 1 mm. Despite
 153 the low number of cycles that these specimens were subjected to, it was also deemed
 154 desirable to avoid using specimens with a rectangular or square cross-section. How-
 155 ever, to limit the degree of machining-induced stress, wire EDM was used to cut
 156 an approximate hexadecagon cross-section resulting in a near-circular profile with a
 157 nominal area of $0.796\ \text{mm}^2$. Due to their size and technical limitations, the specimens
 158 were not polished prior to being tested. Manufacturing tolerances led to deviations
 159 both in terms of the cross-sectional area and shape. This is highlighted in Fig. 3
 160 with the nominal cross-section represented by the dashed lines overlaid upon the

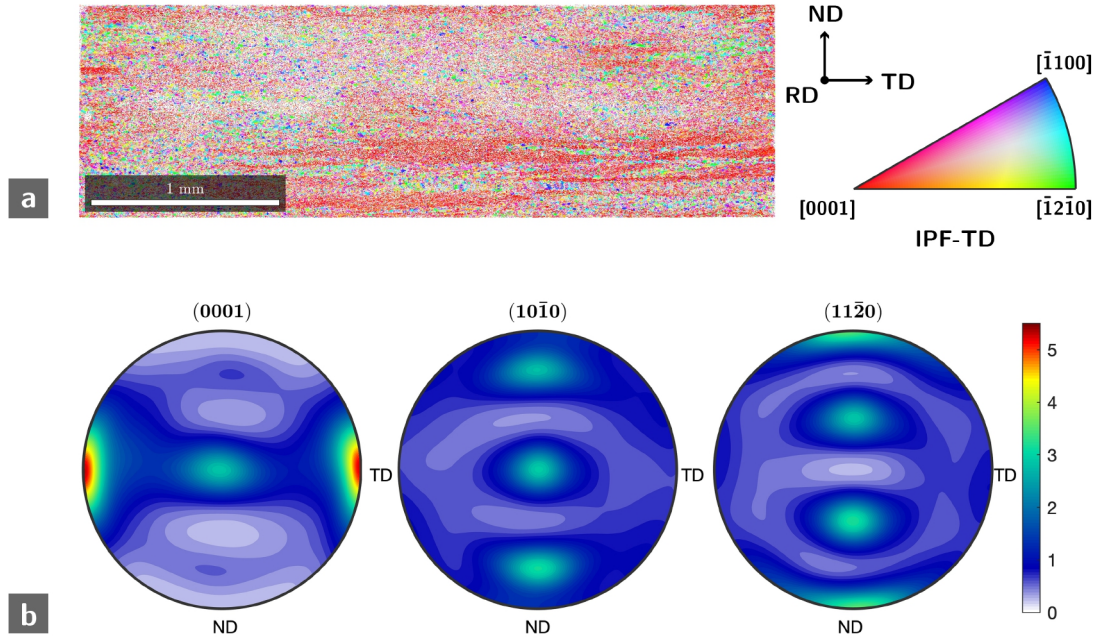


Figure 2: (a) Orientation map of a large area EBSD scan of the bulk material. Local lattice orientations are coloured with respect to the TD axis. (b) Orientation pole figures associated with the same scan (out of plane axis aligns with RD). The majority of basal normal vectors are aligned with the TD and significantly fewer fall in the band between TD and RD.

161 EBSD orientation maps of the true cross-sections.

162 Fig. 3 shows the EBSD orientation maps of the cross-sections of the specimens tested
 163 under dwell cyclic loading conditions. In Fig. 3, the orientation colouring is relative
 164 to the loading direction (LD), which is out of plane and better illustrates the strong
 165 texture differences along the loading axis. Both 45° and TD dwell specimens have a
 166 texture that is visually relatively consistent with the bulk material but have texture
 167 indices (TIs) of 2.91 and 3.62, respectively, which are both higher than the bulk TI
 168 of 2.21. While the RD specimen appears slightly more uneven, with macrozones
 169 concentrated in the top half of the cross-section examined, it has a TI of 2.79 which
 170 is closer to the bulk than the other specimens.

171 2.2. Experimental Method

172 The in-situ experiments were carried out at the F2 station of the Cornell High Energy
 173 Synchrotron Source (CHESS). A BOSE ElectroForce 3300 (3 kN) electromechanical
 174 testing frame was used to apply the load. The target stress was set to 750 MPa,
 175 approximately 85% of a representative yield stress of 880 MPa. No adjustment was
 176 applied to account for inherent differences in the orientation-specific yield stress
 177 or the small manufacturing error which resulted in slightly higher than nominal
 178 cross-sectional areas. Trapezoidal loading waves were used for both load-controlled,
 179 non-dwell and dwell experiments, using 1 s loading and unloading steps. A 0.2 s hold
 180 was added at the peak load for non-dwell experiments to capture the peak strain
 181 during each cycle. The dwell period was 120 s and both sets of experiments were
 182 terminated after 100 cycles.

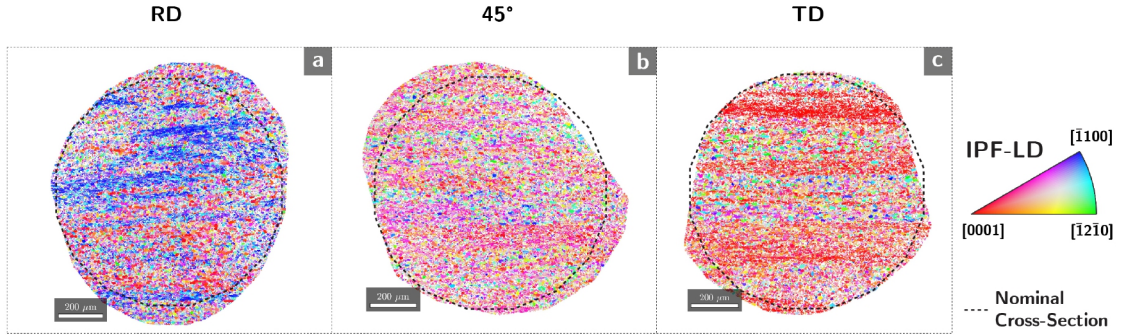


Figure 3: Electron backscatter diffraction images of the cross-sections of the dwell synchrotron specimens. The out of plane axis (LD) coincides with the loading axis. (a) The RD specimen exhibits stronger texture and MTR incidence in the top half with primarily prismatic planes being parallel to the cross-section. (b) Shows the 45° specimen and more uniform distribution of the strong textured regions that are most favourably oriented for basal slip. (c) The TD specimen also displays a macrozone distribution consistent with bulk, with a large proportion of the cross-section being taken up by basal planes being parallel to it in well-defined MTRs.

183 Two Dexela 2923 flat panel detectors capable of capturing images at a frequency
 184 up to 16 Hz and with a 74.8 μm pixel pitch, were used. The energy of the X-ray
 185 beam used was 61.332 keV and the effective beam size was 1 mm x 1 mm. The
 186 sample to detector distance was 795 mm and each detector panel covered an area
 187 of 3888 x 3072 pixels. Calibration was carried out using a CeO₂ standard and the
 188 beam attenuation was adjusted for each specimen. The acquisition rate for the non-
 189 dwell experiments was set to 10 Hz and for the dwell experiments to 1 Hz, which
 190 resulted in 2200 frames for the former and 12200 frames for the latter. Images of
 191 the specimens for the purposes of carrying out DIC were also captured at the same
 192 respective frequencies. DIC images were processed using GOM Correlate [32] to
 193 obtain macroscopic strain data. The experimental setup can be seen schematically
 194 in Fig. 4, including the 15° azimuthal region along the loading axis (bin).

195 2.3. Diffraction Data Processing

196 Post-processing the diffraction images was done using custom Python scripts util-
 197 ising the HEXRD software package [33]. The data was processed for each detector
 198 panel separately as the area captured does not overlap, as shown in Fig. 4. Diffraction
 199 occurs when Bragg's law is satisfied; the crystallographic plane (lattice) spacing
 200 d , beam wavelength λ and diffraction angle θ are related such that:

$$\lambda = 2d \sin \theta. \quad (1)$$

201 For the analysis, peaks from families of lattice planes with diffraction angles 2θ less
 202 than 8° were used. Table 2 provides the diffraction angles for the α and β diffraction
 203 peaks.

204 Obtaining the peak positions used for elastic lattice strain determination throughout
 205 the experiment is performed by first mapping azimuthal regions of intensity on the
 206 detector to a polar coordinate system where the radial coordinate is 2θ and the

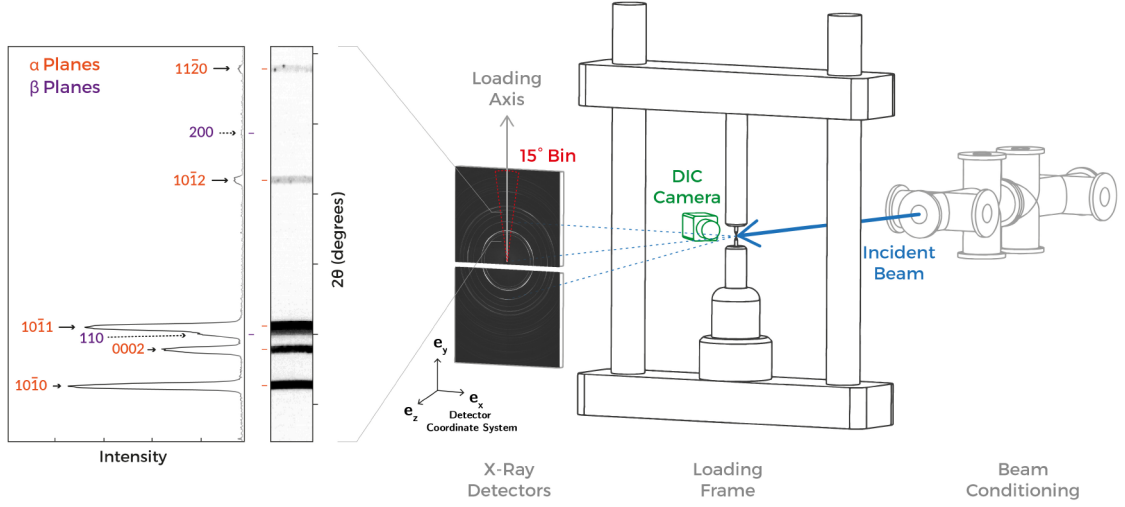


Figure 4: Simplified schematic of the experimental setup including an overview of all considered diffraction rings.

Table 2: 2θ values and their respective hkl s.

hkl	$\{10\bar{1}0\}$	$\{0002\}$	$\{110\}$	$\{01\bar{1}1\}$	$\{01\bar{1}2\}$	$\{020\}$	$\{11\bar{2}0\}$
Phase	α	α	β	α	α	β	α
2θ	4.59	4.96	5.14	5.21	6.76	7.27	7.95

207 angular coordinate is denoted η . Next, the intensity data is azimuthally integrated
 208 over a fine grid. The peak positions and the integrated intensity are calculated
 209 for all peaks of interest. The respective integrated intensity values are numerically
 210 calculated using the Simpson rule. Peak positions were determined using a similar
 211 numerical procedure for the calculation of the centre of mass of the peak. An
 212 arbitrary threshold was set to exclude any remaining noise or extremely weak peaks
 213 that would not produce reliable results. For the remaining intensities, the centre of
 214 mass and the full width at half maximum of each peak was calculated.

215 The rest of the process can be effectively simplified to peak tracking, with the posi-
 216 tion of the peak being registered in a 2θ reference for each frame. Finally, the lattice
 217 strain ε_i^{hkl} for a given frame i and a given family of lattice planes hkl is calculated
 218 as:

$$\varepsilon_i^{hkl} = \frac{d_i^{hkl} - d_0^{hkl}}{d_0^{hkl}} = \frac{\sin(\theta_0^{hkl})}{\sin(\theta_i^{hkl})} - 1. \quad (2)$$

219 Data from top and bottom detector panels, as indicated in Fig. 4, are processed and
 220 averaged to produce the final lattice strain values with the exception of the 45° dwell
 221 experiment where the top panel did not properly trigger, so lattice strains from only
 222 the bottom panel were used.

223 To determine the most appropriate size of the azimuth bin to be considered in the
 224 analysis, the data was processed using 10°, 15°, 20° and 30° bins along the loading
 225 axis. Due to the texture in the UD plate, bin sizes of less than 10° were avoided as
 226 the reduction in the number of grains taken into account both increased the noise
 227 and reduced the statistical accuracy of the results. Bin sizes of 15°, 20°, and 30°
 228 resulted in comparatively less noise. However, for the case of {0002} and {11 $\bar{2}$ 0},
 229 the 20° and 30° bins were found to result in overly smooth changes in lattice strain
 230 at the beginning of the experiment and as such, the 15° bin was chosen.

231 An energy (wavelength) correction was also applied to the data to account for small
 232 fluctuations in the beam energy. Incoming beam energy was monitored throughout
 233 the experiment by intermittent measurements of absorption through a Yb foil. Even
 234 though this correction was applied to reduce noise, it made little difference to the
 235 processed data with the average offset being 2.05×10^{-5} across all experiments.

236 A procedure was also developed to estimate uncertainty in the lattice strain mea-
 237 surements. A primary uncertainty (e) related to the signal-to-noise ratio for each
 238 peak was calculated using the first 10 frames captured before the specimens were
 239 loaded. In the absence of any load, the measured deviation from zero strain was more
 240 pronounced for some orientations and planes but in all cases reflected the primary
 241 source of uncertainty. This uncertainty is calculated as:

$$e = \frac{1}{n} \sum_{i=1}^n |\varepsilon_i| \quad (3)$$

242 where $n = 10$ is the total number of idle frames. For improved clarity regarding the
 243 strain evolution trends with increased cycling, the average value per cycle peak for
 244 each hkl is calculated.

245 *2.4. Applied Macroscopic Stress*

246 The observed variation of the cross-sectional area, as shown in Fig. 3, necessitated
 247 a review of the induced macroscopic stresses to aid interpretation of the various
 248 experiments, as well as to confirm that no specimen was loaded too close to the
 249 macroscopic yield stress. To this end, the specimens were sectioned at their nominal
 250 gauge section and lightly polished to deburr and reveal the sharp cross-sectional
 251 outline. The cross-sections were scanned in 2D using an Alicona InfiniteFocus pro-
 252 filometer fitted with a 10x lens.

253 Each section was scanned twice with a polygon manually fitted around its outline
 254 and used to compare each specimen's average cross-sectional area relative to the
 255 nominal value. In addition, the applied load of 600 N was divided by the measured
 256 area and compared to the orientation-dependent yield stresses σ_y .

257 These yield strengths were determined through separate small scale tensile tests
 258 performed after the X-ray measurements using a Deben 5 kN stage from the same
 259 batch of material. For these, the crosshead displacement rate was set to 1 mm min^{-1} ,
 260 which corresponded to an average strain rate of $1.28 \times 10^{-2} \text{ s}^{-1}$ in the elastic regime

261 and $1.89 \times 10^{-1} \text{ s}^{-1}$ in the plastic regime. The 0.2% proof stresses were estimated to
 262 be 932 MPa for RD, 904 MPa for 45° and 990 MPa for TD.

263 It is worth noting that the scans were taken after the experiments and hence the
 264 cross-sectional area is that of the deformed state. However, due to the loading
 265 being limited to within the macroscopic elastic regime and the relatively small to
 266 near-zero accumulation of strain, it is reasonable to assume a negligible change in
 267 cross-sectional area. The area data, yield strengths, true applied stress and ratio of
 268 applied stress to yield strengths are summarised in Table 3.

Table 3: Overview of estimated macroscopic stresses. The area has been measured after the experiments and σ_y is the orientation-dependent yield stress determined by separate small scale tests. σ is the actual macroscopic stress determined from the true cross-sectional area.

Specimen	Area (mm ²)	σ_y (MPa)	σ (MPa)	σ/σ_y
RD/Non-Dwell	0.8429	932.4	711.9	0.764
RD/Dwell	0.9545	932.4	628.6	0.674
45°/Non-Dwell	0.8478	903.8	707.7	0.783
45°/Dwell	0.9274	903.8	646.9	0.716
TD/Non-Dwell	0.7850	989.8	764.3	0.772
TD/Dwell	0.7805	989.8	768.7	0.776

269 Table 3 reveals that the average actual loading is estimated to be 74.8% (standard
 270 deviation = 3.9%) of the orientation-specific yield, which despite the deviation from
 271 the original target of 85% remains satisfactory for a low-cycle dwell experiment. For
 272 Ti64, it has been shown that peak stresses that exceed 85% of the yield accumulate
 273 significantly more plastic strain due to cold creep, whereas reduced plastic strain
 274 accumulation is observed for values in the 60-80% range [34]. Taking into account the
 275 particularly tight tolerances for manufacturing and sectioning, an accumulation of
 276 error is to be expected. Crucially, however, it is important to note that all specimens
 277 were loaded to a sufficiently sub-yield stress such that the variance should have a
 278 negligible effect on the onset of plastic deformation.

279 2.5. Estimation of Orientation-Dependent Stresses

280 To further analyse load redistribution among the sets of planes that diffract along
 281 the loading axis, it is helpful to estimate the average stresses along that direction
 282 from lattice strain measurements using directional moduli. Hooke's law provides the
 283 relationship between stress σ_{hkl} , elastic lattice strain ε_{hkl} and elastic moduli E_{hkl} as:

$$\sigma_{hkl} = E_{hkl}\varepsilon_{hkl}. \quad (4)$$

284 The elastic constants used in the calculation of the directional moduli are taken
 285 from Wielewski et al [28] and are (in Voigt notation) $C_{11} = 169 \text{ GPa}$, $C_{13} = 62 \text{ GPa}$,
 286 $C_{12} = 89.0 \text{ GPa}$, $C_{44} = 43 \text{ GPa}$, $C_{33} = 196 \text{ GPa}$. The calculated directional moduli
 287 are summarised in Table 4.

Table 4: Directional moduli for the sets of grains contributing to the diffraction peaks considered.

hkl	$\{10\bar{1}0\}$	$\{0002\}$	$\{01\bar{1}1\}$	$\{01\bar{1}2\}$	$\{11\bar{2}0\}$
Modulus (GPa)	104.4	143.3	109.0	118.5	104.4

3. Results

3.1. Macroscopic Strain Accumulation

The macroscopic strain evolution obtained via DIC is useful for assessing whether the addition of dwell had the expected effect on strain accumulation. Fig. 5 (a) shows the accumulation of macroscopic strain $\Delta\varepsilon_{\text{macroscopic}}$ for specimens loaded under normal cyclic loading conditions, while Fig. 5 (b) shows the accumulation of macroscopic strain in specimens that were cycled with a dwell period at peak load. It is noted that the lighting conditions at the X-ray station were not ideal for DIC measurements leading to an inability to reconstruct macroscopic strain at all points through the non-dwell cyclic loading. Nonetheless, from the instances captured, it can be established that there is no notable accumulation of strain in the absence of dwell periods in any of the specimens.

In the macroscopic strain data for the specimens that were cycled with dwell periods seen in Fig. 5 (b), the 45° specimen stands out as having accumulated a relatively significant amount of strain, reaching up to 4.6×10^{-3} . For both RD and TD specimens, the rate of strain accumulation quickly drops in the first five cycles and becomes fairly stable from the twentieth cycle onwards, ultimately not exceeding 1.0×10^{-3} by the hundredth cycle.

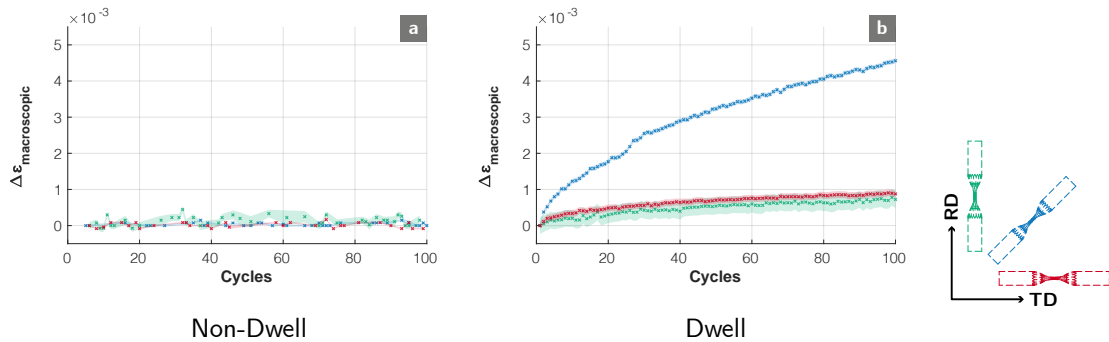


Figure 5: Evolution of peak macroscopic strain (DIC) over 100 cycles for specimens along all tested orientations. (a) shows all specimens subjected to normal (non-dwell) cyclic loading not accumulating any strain. (b) shows the evolution for specimens subjected to a 120 s dwell in each cycle, revealing notable accumulation for the 45° specimen.

3.2. Lattice Strain Evolution

The lattice strain measurements at peak load during each cycle were averaged for all respective families of lattice planes to explore the changes in load distribution with the increasing cycle count. To emphasise the changes in these lattice strains

at peak, we define a change of strain $\Delta\varepsilon^{hkl}$ from a given cycle i relative to the first cycle by:

$$\Delta\varepsilon^{hkl} = \varepsilon_i^{hkl} - \varepsilon_0^{hkl}, \quad (5)$$

where ε_0^{hkl} is the respective mean value of the first cycle peak. The calculated lattice strain evolution for all families of α lattice planes from all specimens is shown in Fig. 6. It should be noted that intensities from two families of lattice planes from the β phase were present in the diffraction angle range analysed. However, uncertainties in the positions of these peaks were high due to low intensity and the proximity to α peaks, and as such, they are not presented. For all other measurements, the shaded areas represent the uncertainty as defined in Equation 3. A persistent, elevated uncertainty was identified for the $\{01\bar{1}1\}$ family, predominantly in the specimens along the RD and TD, as the texture and relative orientations led to low intensities along the loading axis.

3.2.1. Non-Dwell Experiments

Initially, the response in the absence of dwell appears fairly consistent between orientations and slip planes, as shown in Fig. 6 (a)-(e) with few and subtle differences noted. Along the RD and TD, cyclic hardening of the planes diffracting along the loading direction has been observed with magnitudes ranging from $-2 \times 10^{-4} \leq \Delta\varepsilon \leq -4 \times 10^{-5}$ for all families except $\{01\bar{1}2\}$ where the RD response revealed a larger drop in strain to a magnitude of $\Delta\varepsilon \approx -3.3 \times 10^{-4}$. The basal and prismatic response of the RD and TD specimens consists of a relatively small but sharp drop in strain between the first and second cycle peaks, showing little evolution for the remainder of the experiment. It is also worth noting that despite the sharp initial drop of $\{11\bar{2}0\}$ strain along the RD, it then increases gradually for the remaining cycles (Fig. 6 (e)). Some subtle differences in the elastic response of the 45° experiment are noted; only the $\{01\bar{1}2\}$ system exhibits a clear decrease in lattice strain with applied cycles, measuring a $\Delta\varepsilon \approx -2.2 \times 10^{-4}$ at the end of the experiment. In the 45° experiment, both $\{0002\}$ and $\{01\bar{1}1\}$ lattice strains show a minimal deviation from zero not exceeding $\pm 1.0 \times 10^{-4}$ at any point during the experiment. The prismatic response reveals a slight increase which reached a magnitude of up to 1.0×10^{-4} (Fig. 6 (a)).

In the non-dwell specimens, it has been highlighted that some lattice strains show a sharp drop between the first and second cycles. This is most likely attributed to redistribution of residual stresses, which stem primarily from the manufacturing process, but can also be affected by the local texture [35]. This is more pronounced for the RD results and, despite this not being reflected on the macroscopic response, it is relatively significant for systems such as $\{0002\}$ and $\{01\bar{1}0\}$ for which the evolution between first and third cycle is larger than between the third and hundredth cycle.

3.2.2. Dwell Experiments

More pronounced lattice strain evolution is displayed with the addition of dwell, as shown in Fig. 6 (f-j). As anticipated, the TD specimen appears to be the least affected by the hold, as a cyclic drop is also observed for $\{01\bar{1}0\}$, $\{01\bar{1}1\}$ and $\{01\bar{1}2\}$ strains, whereas no notable evolution has been observed for the $\{11\bar{2}0\}$ family. For

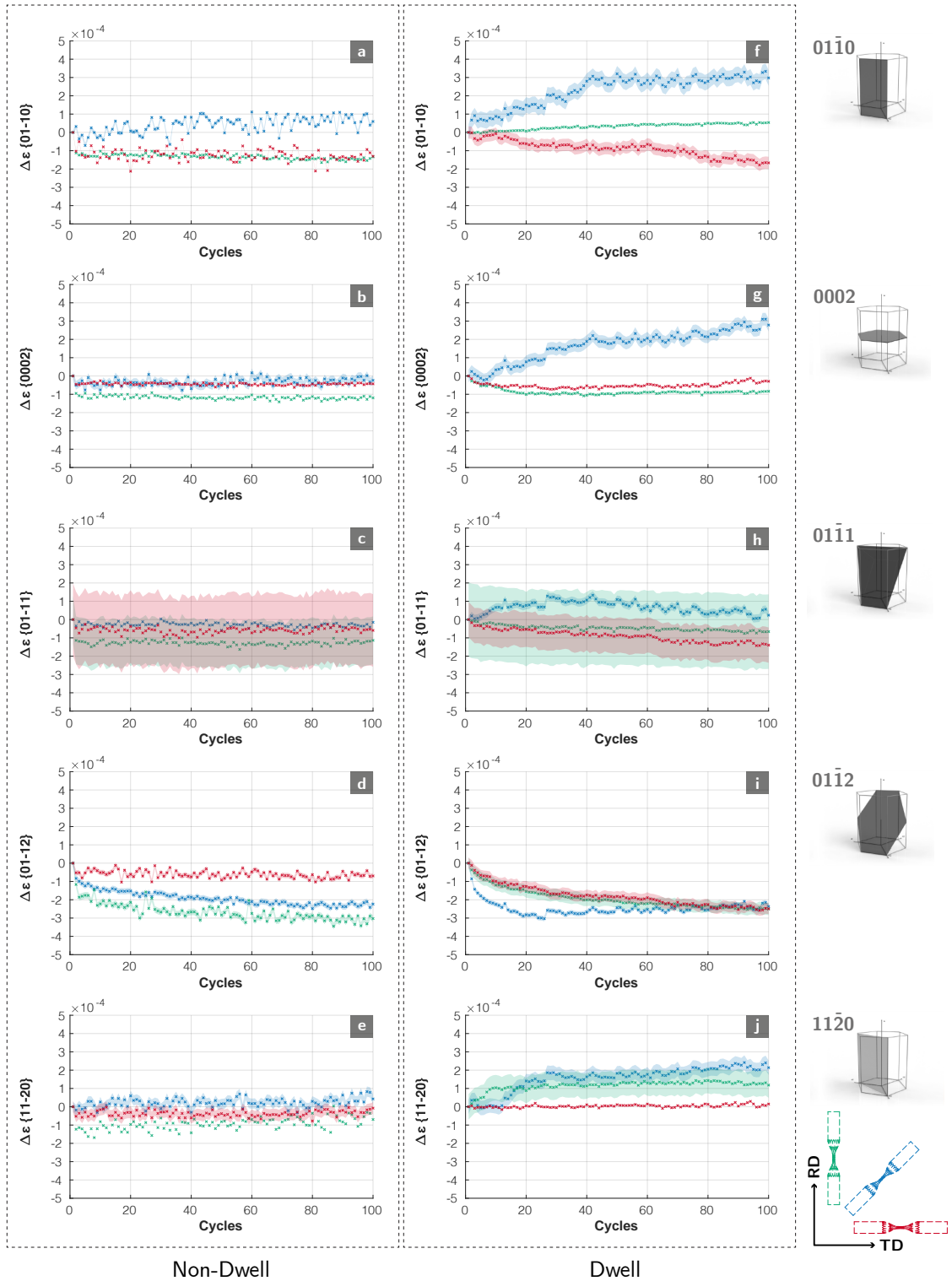


Figure 6: Overview of the lattice strain evolution for the $\{01\bar{1}0\}$ family for non-dwell (a) and dwell (f) loading, for the $\{0002\}$ family for non-dwell (b) and dwell (g) loading, for the $\{01\bar{1}1\}$ family for non-dwell (c) and dwell (h) loading, for the $\{01\bar{1}2\}$ family for non-dwell (d) and dwell (i) loading, for the $\{11\bar{2}0\}$ family for non-dwell (e) and dwell (j) loading.

348 this orientation, the basal response should be noted as it shows a gradual drop in
349 $\{0002\}$ strain over the first ~ 10 cycles, remaining mostly unchanged over the next
350 ~ 60 cycles and finally showing a small increase in measured lattice strain as the
351 completion of 100 cycles approached.

352 The RD specimen reveals strong similarities with TD for basal and pyramidal strains,
353 but with key differences in the prismatic families. Even though the magnitude
354 of strain accumulation in $\{01\bar{1}0\}$ is small ($\Delta\epsilon < 6 \times 10^{-5}$), it does represent a
355 switch to cyclic increase when compared to the non-dwell experiment along the
356 same orientation. The $\{11\bar{2}0\}$ strain for the same specimen is much more scattered
357 but exhibits a small increase in the first ~ 8 cycles before plateauing at a magnitude
358 of $\Delta\epsilon < 1.5 \times 10^{-4}$.

359 The 45° specimen under dwell conditions reveals the largest strain increase between
360 dwell experiments with the most prominent ones exhibiting, on average, a $\Delta\epsilon \approx$
361 3.1×10^{-4} for $\{01\bar{1}0\}$ and $\Delta\epsilon \approx 3.0 \times 10^{-4}$ for $\{0002\}$ at the end of the experiment.
362 A closer inspection reveals that in most cases the trends change within or at ~ 40
363 cycles; within the first 10 cycles basal strain slightly fluctuates and drops before
364 stabilising to a positive gradient for the remainder of the dwell experiment. The
365 strain evolution for $\{01\bar{1}1\}$ shows a multitude of gradient changes as it increases
366 more sharply in the first 10 cycles, plateaus irregularly for the following ~ 45 cycles
367 and then begins gently dropping for the remainder of the experiment. Similarly for
368 the $\{01\bar{1}2\}$ family of lattice planes, the 45° specimen records the largest drop in the
369 first 25 cycles which amounts to a $\Delta\epsilon \approx -3.0 \times 10^{-4}$; from which point the gradient
370 becomes slightly positive. Finally, the second-order prismatic lattice strain reveals
371 a moderate overall increase of $\Delta\epsilon \approx 2 \times 10^{-4}$ but a decrease is observed over the
372 first ~ 10 cycles followed by an increase over the subsequent ~ 15 cycles before
373 plateauing.

374 *3.3. Orientation-Dependent Stresses*

375 To further characterise load redistribution during the cyclic loading experiments,
376 changes in stress were calculated from the changes in lattice strains presented above.
377 The estimated orientation-dependent stresses for all experiments are shown in Fig. 7.
378 The relative change in lattice strain is multiplied by the previously calculated direc-
379 tional moduli resulting in a $\Delta\sigma$ that is relative to the first cycle peak. Complimentary
380 uncertainties are those calculated for lattice strains, also scaled by the directional
381 moduli. We emphasise here that these stresses assume a uniaxial stress state in all
382 the grains which is well known to not be true. However, these changes in stress
383 do provide some new insights into load redistribution among sets of grains in the
384 volume probed.

385 Even though the key trends in lattice strain evolution are also reflected in the $\Delta\sigma$
386 plots, the scaling effect of the respective moduli aids in the identification of further
387 differences between the orientations and the cyclic loading with and without a dwell
388 step.

389 It is apparent that the addition of dwell periods has a different effect on the evolu-
390 tion of stresses within the RD and TD specimens which macroscopically exhibited

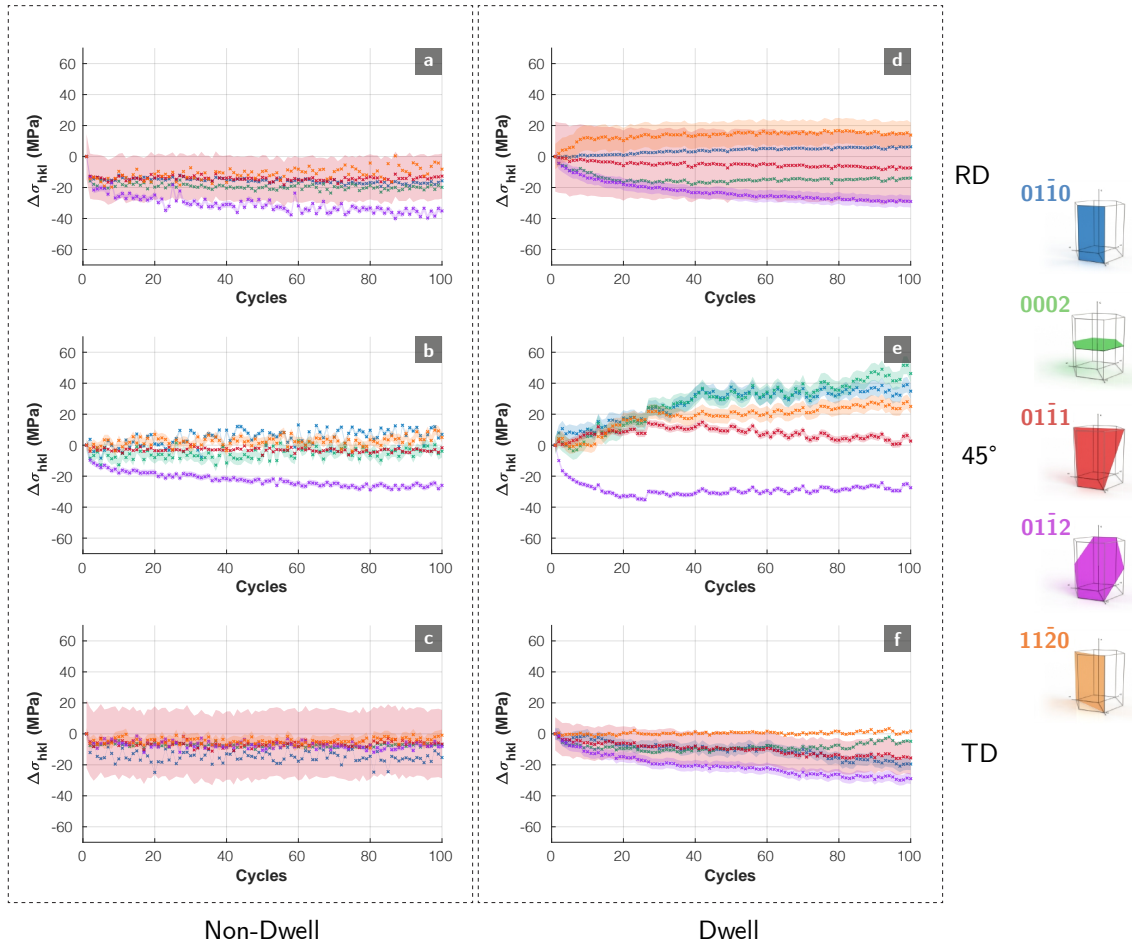


Figure 7: Overview of the calculated stress evolution based on lattice strains for the RD specimens subjected to non-dwell (a) and dwell (d) loading, for the 45° specimens subjected to non-dwell (b) and dwell (e) loading, for the TD specimens subjected to non-dwell (c) and dwell (f) loading.

391 a very similar and limited dwell debit after 100 cycles. In the most dwell-insensitive
 392 orientation, TD, the $\{01\bar{1}0\}$, $\{01\bar{1}1\}$ and $\{01\bar{1}2\}$ stresses closely evolve along a nega-
 393 tive trajectory, but $\{0002\}$ stress reflect a flip from a negative to a positive gradient
 394 after approximately 70 cycles (Fig. 7 (f)).

395 The crystallographic conditions present in the 45° non-dwell specimen have led to
 396 a different stress evolution under dwell loading. However, it can be seen that the
 397 sets of grains contributing to $\{0002\}$ and $\{01\bar{1}0\}$ peaks, with their c axes at 0° and
 398 90° from the loading axis respectively, experience the largest increase in stress which
 399 momentarily exceeds 51.5 MPa (Fig. 7 (e)). The 45° specimen subjected to dwell
 400 reveals a number of changes in evolution in the first ~ 25 cycles, but thereafter,
 401 there is a divergence between $\{01\bar{1}0\}$, $\{0002\}$ and $\{11\bar{2}0\}$ which keep increasing and
 402 $\{01\bar{1}1\}$ which keeps decreasing.

4. Discussion

The macroscopic and lattice strain response of UD-rolled Ti64 has been captured in-situ under cyclic (non-dwell) and 120 s dwell conditions over 100 cycles. The form of the alloy chosen presents a crystallographically intriguing case, which allowed MTRs to be subjected to loading at different orientations relative to the dominant macrozone orientation by extracting specimens along the plate’s RD, TD and 45° between the two. These measurements reveal subtle but key differences arising from the combination of dwell and orientations effects related to the MTRs. This study complements a wide range of research on titanium alloys for aerospace applications [8] and relatively recent incidents of in-service components being heavily affected by the material’s texture [36] highlight the need for continuous and in-depth reviews of our understanding of key deformation mechanisms.

4.1. Macroscopic Response

Macroscopic strain measurements obtained via DIC reveal no appreciable accumulation of strain in any of the tested orientations over the course of the non-dwell experiments. When subjected to dwell, the macroscopic response changes to reveal limited evolution for both RD and TD specimens, which did not exceed 1.0×10^{-3} , but a significant accumulation of 4.6×10^{-3} was recorded for the 45° specimen. The rate of strain accumulation in the 45° specimen declines quickly in the first 40 cycles but starts stabilising onwards to an average rate of 2.78×10^{-5} per cycle.

To further aid in the interpretation of this result prior to considering the grain-scale response, EBSD data has been used to calculate the maximum basal and prismatic Schmid factor (SF) across the cross-sections of the dwell specimens, which are shown in Fig. 8. Even though the maps represent only a slice of the sampled volume for diffraction, they offer a representative view of the sites likely to experience the highest resolved shear stresses during loading. As shown in Fig. 2, the rolling process causes many grains to preferentially align their c axis with the TD and, as such, it is sensible that the 45° specimen presents the most favourable case for basal slip. In addition, Fig. 8(b) reveals a high basal SF distribution over almost the entirety of the cross-section, suggesting a synergy of grains that are part of TD-oriented macrozones and some grains in the weaker-textured regions with equally high SFs (primarily those that are oriented along the RD). The same specimen shows a moderate distribution of prismatic SFs in Fig. 8(e), suggesting that prismatic slip could also be activated particularly after grains favourable for basal slip have been significantly deformed and lost their load-carrying capability. The second-highest average SF has been observed for the prismatic family in the RD specimen (Fig. 8(d)), which did not manifest itself in a significant accumulation of macroscopic strain. The TD specimen also shows a moderate basal SF distribution (Fig. 8(c)), which is largely counteracted by the macrozones displaying extremely low prismatic SFs (Fig. 8(f)).

4.2. Grain-scale Response

The peak lattice strain in cyclic (non-dwell) fatigue showed little evolution that didn’t exceed $|\Delta\varepsilon| \leq 1.5 \times 10^{-4}$ over 100 cycles for all specimens, with the exception

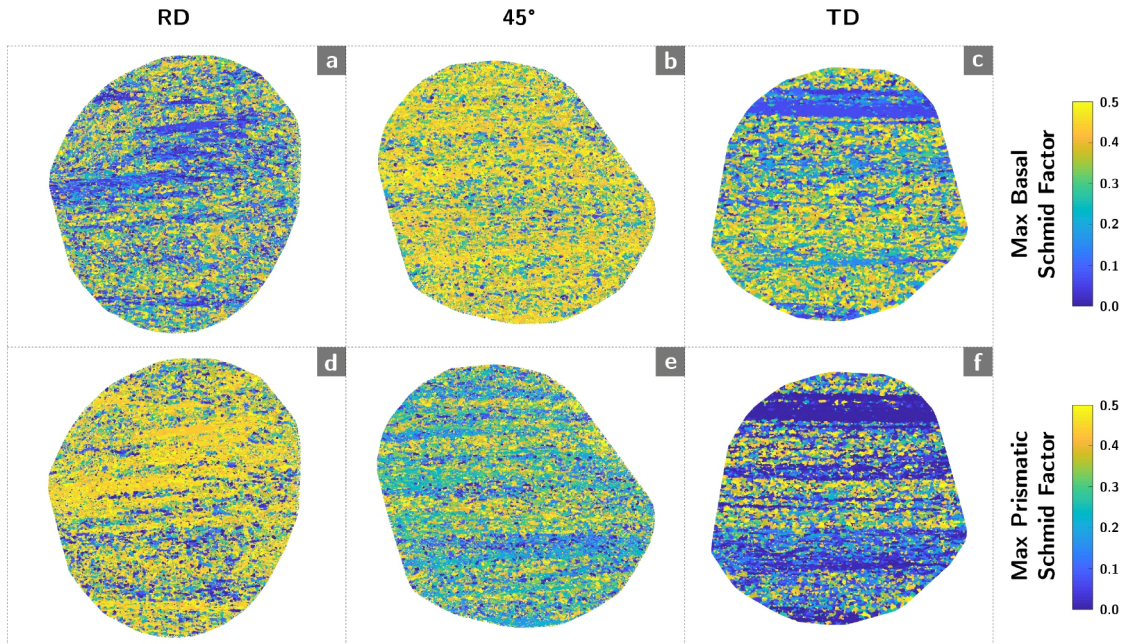


Figure 8: Overview of the calculated Schmid factors based on the EBSD data for the basal family and the RD (a), 45° (b) and TD (c) specimens, as well as the prismatic family and the RD (d), 45° (e) and TD (f) specimens.

446 of the $\{01\bar{1}2\}$ family which recorded a drop of up to $\Delta\varepsilon \approx -3.4 \times 10^{-4}$ in the
 447 RD specimen. This apparent cyclic hardening behaviour might seem consistent at
 448 first since it concerns a family with a high critical resolved shear stress relative to
 449 basal or prismatic, but at this point, it is important to make a distinction between
 450 easy/hard planes to slip and where each of these measurements come from. For
 451 example, it is known that a grain with its c axis 45° away from the loading axis,
 452 experiences higher basal stress than in any other orientation, but such a grain would
 453 also diffract an incoming beam 45° away from the loading axis on the diffraction ring.
 454 The measurements displayed in Fig. 6 stem from diffracted points within 7.5° of the
 455 loading axis. This is consistent with the azimuth bin chosen for the analysis, as the
 456 15° segment symmetrically stretches over both sides of the loading axis corresponding
 457 to a deviation of up to 7.5° in any direction. As such, a ‘soft’ grain for basal slip
 458 does not contribute its basal lattice strain response along the loading axis, but 45°
 459 away from it. However, in the same grain, the $\{01\bar{1}2\}$ plane is approximately 42.4°
 460 away from the basal plane and that would, in fact, diffract within 7.5° of the loading
 461 axis.

462 It is worth noting that the mechanical response from a region of the material wholly
 463 taken up by a macrozone is significantly different from a weak-textured region, as
 464 the former behaves more like a single grain. In production materials, both of these
 465 distinct regions exist, but the inherent differences in mechanical properties lead to
 466 strain localisation [37, 38]. Regarding the interpretation of the diffraction data, this
 467 means that virtually all grains with an orientation along the dominant macrozone
 468 orientation (TD) will only contribute measurements to one family, whereas the rest

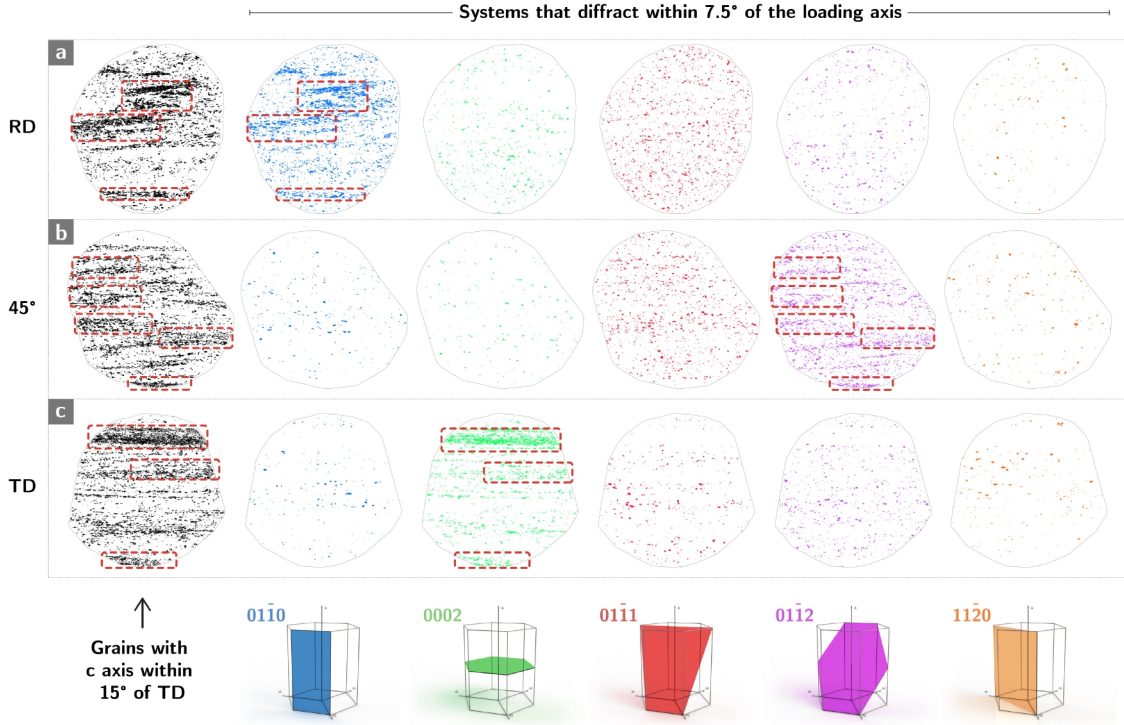


Figure 9: Fibres of orientations calculated from the EBSD data, highlighting which grains have c axes within 15° of the dominant macrozone orientation, TD. On their right, EBSD data for all grains is used to highlight the respective crystal planes that would diffract the incident beam within 7.5° of the loading axis for the RD (a), 45° (b) and TD (c) specimens. The dotted red boxes highlight some of the more prominent MTRs and the corresponding families they diffract from.

469 will be measured only from the weaker-textured regions. More crucially, the family
 470 that macrozones contribute lattice measurements to is different in all three cases
 471 examined. To further clarify this, the EBSD data has been used to visualise orien-
 472 tation fibres contributing to diffraction along the loading axis are shown in Fig. 9.
 473 On the left of the figure, a threshold map highlights grains with c axes along the
 474 TD, while the plots on the right highlight all respective α planes with normal plane
 475 vectors pointing within 7.5° of the loading axis.

476 Even though Fig. 9(b) confirms that grains within the 45° specimen's macrozones
 477 predominantly contribute to the $\{01\bar{1}2\}$ measurement along the loading axis, closer
 478 inspection in conjunction with Fig. 8(a) and (c) reveals that high basal SF areas are
 479 primarily represented by $\{01\bar{1}2\}$ measurements for the RD and TD specimens too.
 480 In essence, the drop in $\{01\bar{1}2\}$ family strain for both RD and 45° specimens is most
 481 likely a reflection of basal slip taking place in those grains. The stress redistribution
 482 causes an ever-reducing load to be applied to those grains and, subsequently, being
 483 resolved onto $\{01\bar{1}2\}$. Even in the absence of dwell, the role of the β phase in early
 484 deformation should not be discounted [39–41] but this could not be examined in this
 485 case for either $\{110\}$ or $\{020\}$.

486 The addition of the dwell period had an observable effect on all specimens at the lat-
 487 tice level. The RD specimen, despite having high prismatic SFs (Fig. 8(d)) and with

488 those same grains contributing to the measured prismatic response (Fig. 9(a)) only
 489 recorded a marginal increase in $\{01\bar{1}0\}$ family strain. High stresses in macrozones
 490 with predominantly prismatic orientations have been shown to interact with basal
 491 textures that eventually lead to cracks forming at their interface [42]. In this case,
 492 the prismatic macrozones did not manage to sufficiently interact with favourable
 493 basal planes, possibly due to the absence of enough favourably oriented neighbours.
 494 Even though the TD specimen subjected to dwell loading shows a drop in $\{01\bar{1}0\}$,
 495 $\{01\bar{1}1\}$ and $\{01\bar{1}2\}$ family strain, the overall behaviour is dominated by macrozones
 496 which are oriented unfavourably for both basal and prismatic slip. Those macro-
 497 zones only directly contribute measurements to the $\{0002\}$ family, which shows little
 498 evolution in support of the previous statement. From both Fig. 8(f) and Fig. 9(c), it
 499 can be suggested that the prismatic response for the TD specimen which showed a
 500 gradual drop over the course of the experiment stems from relatively fewer grains in
 501 the weak-textured region which also has high prismatic SF and indicates that some
 502 prismatic slip may have taken place but at a very limited scale and fully surrounded
 503 by much stronger-textured regions.

504 In the particular case of the 45° specimen, the anticipated response is captured
 505 macroscopically, but as Fig. 9(b) confirms, both the $\{01\bar{1}0\}$ and $\{0002\}$ family re-
 506 sponses stem from grains in the weak-textured regions. Even though the lattice
 507 strain response might seem slightly more erratic in the first ~ 25 cycles it could
 508 not be attributed to an external factor (such as loss of grip), as no strong discon-
 509 tinuities are present in the macroscopic strain (DIC) data. As such, the observed
 510 response most likely reflects the complex stress redistribution taking place during
 511 the early deformation cycles. As noted, the macrozones in this specimen have only
 512 contributed to the $\{01\bar{1}2\}$ response.

513 However, if one takes into account the relative orientation relationships discussed
 514 so far and consults Figs. 6(f), (g), and (i) in tandem, a chain of events can be pos-
 515 tulated. In this specimen, the $\{01\bar{1}2\}$ family exhibits the fastest strain drop in 25
 516 cycles among all families and all experiments while also being an indirect reflection
 517 of the conditions within the macrozones (Fig. 9(b)) and some of the other areas
 518 with very high basal SF (Fig. 8(b)). While this is taking place, both the $\{01\bar{1}0\}$
 519 and $\{0002\}$ family strains increase, at a slightly faster rate for the prismatic fam-
 520 ily which plateaus after the fortieth cycle, while the basal strain keeps increasing
 521 over the course of the experiment. Even though both $\{01\bar{1}0\}$ and $\{0002\}$ measure-
 522 ments come from weak-textured regions, only the former displays very high SFs
 523 which could elucidate why the $\{01\bar{1}0\}$ response plateaus after a point. Overall, this
 524 would suggest widespread basal slip in the MTRs from the first few dwell applica-
 525 tions which significantly slows down after approximately thirty cycles, as indicated
 526 by the $\{01\bar{1}2\}$ response. As this is taking place, some of the initial load is being
 527 redistributed to grains with their c axes at $0^\circ \pm 7.5^\circ$ to the loading axis (suggested
 528 through the $\{0002\}$ response) and grains with their c axes at $90^\circ \pm 7.5^\circ$ (suggested
 529 through the $\{01\bar{1}0\}$ response), with one important distinction being that after the
 530 fortieth cycle some wider prismatic slip is also taking place outside the MTRs. This
 531 is in line with findings from recently published work [43] which showed that in cases
 532 where basal slip is active but the load is not sufficiently high to initially activate

533 slip on other planes, the cyclic redistribution of load leads to concurrent slip with
534 the number of active slip systems increasing as the accumulated plastic strain also
535 increased. The $\{0002\}$ family keeps accumulating elastic strain over the course of
536 the experiment and ultimately such grains (with c axis $0^\circ \pm 7.5^\circ$ away from the load-
537 ing axis) become prime candidates for facet formation sites [3]. Even though the
538 dwell mechanism is associated with a plastic strain debit which cannot be captured
539 in these measurements, the indirect shifts in lattice strain gradients that it caused
540 are instrumental in interpreting the mechanism's progression.

541 The utilisation of the captured data along with the approximation of the directional
542 moduli enabled the estimation of the stresses the lattice experienced and highlighted
543 further differences. The 45° specimen exhibits the most rapid change in stress, of
544 $\Delta\sigma \approx -32.7$ MPa for $\{01\bar{1}2\}$ within the first twenty cycles and the largest increase
545 of $\Delta\sigma \approx 51.5$ MPa for the $\{0002\}$ family towards the end of the experiment. The
546 effects of stress redistribution due to residual stresses were more pronounced for
547 some of the non-dwell experiments as the largest change in stress has been observed
548 in the first few cycles. The stress evolution plots also highlight that stresses in the
549 TD specimen subjected to dwell either remained unchanged ($\{11\bar{2}0\}$) or decreased
550 over the course of the experiment. The necessity for balance in stress redistribution,
551 particularly in a specimen with very limited macroscopic evolution, suggests that
552 stress must have increased in either the β phase or in groups of α families that did
553 not diffract near the loading axis or were affected by texture inhomogeneity.

554 *4.3. Size Effects*

555 The occurrence of more or less favourable orientations for slip in macrozones presents
556 further challenges, as their nature is such that they do not pose significant barriers
557 to slip transmission within the MTR itself. Even though some work has been done to
558 examine the progression of deformation mechanisms at the interface of macrozones
559 and weak-textured regions [16, 17, 19], an important consideration, in this case, is
560 that some of the macrozones are not fully or sufficiently enclosed by a weak-textured
561 matrix. A computational study by Liu and Dunne [44] noted that high-aspect
562 macrozones that are fully subsurface are more damaging than those that intersect
563 free surfaces. However, this has not been validated experimentally and based on
564 these experiments, cases where one or more macrozones are found to intersect free
565 surfaces on one or both ends should not be necessarily discounted as less sensitive to
566 dwell fatigue. For this case, the potential for an augmented influence of the MTRs
567 due to their relative volume to the sample volume at gauge has been considered and
568 will be the topic of subsequent investigation.

569 Even though rectangular specimens are known to underperform compared to round
570 ones for some materials, both in tensile tests [45] and under fatigue [46], the choice
571 of the hexadecagon as a feasible cross-section and the manufacturing limitations
572 imposed a sizing error that was difficult to quantify in-situ. Efforts were subsequently
573 made to quantify the variation and this was found to be moderate while confirming
574 that no specimen was overloaded with the average actual stress being determined
575 to be $\sim 74.8\%$ of the orientation-dependent yield stress for UD-rolled Ti64. We
576 note that although there was variation in loading conditions, the dwell specimens all

577 exhibited the most prominent load redistribution with cyclic loading, even though
578 these specimens tended to have lower peak stresses. A peak stress of 60% of the yield
579 is often cited as the lowest necessary for the onset of dwell due to Ti64 undergoing
580 cold creep at these stress levels [47], although this has not been rigorously validated
581 in a cold-dwell fatigue context. Other studies have indicated that dwell effects are
582 considered to become prominent at stresses that are approximately $\geq 75\%$ of the
583 yield stress [2]. In this study, despite the deviation from the target stress level of
584 85%, all specimens were cycled above 60% of the yield stress and all but two were
585 cycled above 75% of the yield stress. The 45° specimen subjected to dwell was one
586 of the two cycled below the 75% yield stress threshold, but showed comparatively
587 large plastic strain accumulation over 100 cycles. This represents a useful result as it
588 captures the dwell evolution at a stress that minimises the risk of other mechanisms,
589 such as plastic ratcheting, playing a significant role in the deformation. The RD
590 specimen subjected to dwell was the only specimen loaded to less than 70% of the
591 yield stress and this loading places it close to the roughly defined threshold of dwell-
592 sensitive stresses. However, the subtle evolution of the lattice response indicates
593 that a certain amount of load redistribution had taken place even in this specimen.
594 These results suggest that dwell fatigue may be operative at relatively low stress
595 levels and should be the subject of further investigation. For future work, it is worth
596 reconsidering the choice of simpler cross-sections or other manufacturing methods.
597 Furthermore, smaller specimens have been known to underperform in dwell fatigue
598 testing; experiments by Song and Hoepfner on IMI 829 support this argument [48],
599 but the ability to probe the entire gauge volume justified the choice of size.

600 5. Conclusions

601 The role of large microtextured regions in Ti64 and dwell fatigue has been examined
602 using high energy X-ray diffraction with the dominant MTR orientation aligned
603 with the loading axis, perpendicular to the loading axis and at 45°. Specimens were
604 subjected to cyclic (non-dwell) and 120s dwell loading for 100 cycles at an average
605 of 74.8% of the orientation-dependent yield stress.

- 606 • In the absence of dwell, the rolled material exhibits near zero or subtly de-
607 clining strains at the grain level over 100 cycles with the dominant macrozone
608 orientation c axis both aligned with the loading axis (TD) and with it lying
609 perpendicular to the loading axis (RD).
- 610 • Despite not showing signs of significant dwell debit macroscopically, both the
611 RD and TD specimens display a different evolution of peak lattice strains under
612 the application of the dwell hold when compared to the respective non-dwell
613 responses.
- 614 • A complex redistribution of stresses at the grain level has been captured within
615 the first 25 cycles of dwell fatigue for the 45° specimen. The uniformly high
616 maximum basal SF distribution across the specimen's cross-section contributed
617 to notable macroscopic strain accumulation which increased by 4.6×10^{-3} over
618 the course of the experiment.

619 • The 45° specimen recorded the quickest drop in $\{01\bar{1}2\}$ family strain, reaching
620 -3.0×10^{-4} in 25 cycles. An examination of the diffraction fibres and the
621 crystallographic conditions of the grains contributing to each family's response
622 indicated that this response stems from grains within the MTR with high
623 basal SFs. In the first 40 cycles, the largest increase in lattice strain has been
624 observed for the $\{01\bar{1}0\}$ and $\{0002\}$ families, which postulated that at least a
625 proportion of the load being redistributed following the early basal slip is being
626 received by grains in the weak-textured matrix with their c axes at $90^\circ \pm 7.5^\circ$
627 and $0^\circ \pm 7.5^\circ$ away from the loading axis respectively.

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636 Conflict of Interest

637 The authors declare that they have no conflicts of interest.

638 Data Availability

639 The raw and processed data required to reproduce these findings cannot be shared
640 at this time as the data also forms part of an ongoing study.

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